U.S. DEPT. COMM./NORA/DAD - DATA SECTION WORK FORM NO.1 DROWF1 FILE FLT ID: 960622 H FM: REF TO: 4 FLT NO: 16-029 BLK IN: ATA: ETD: 2000 20001959 BLK OUT: ATD: 2010 0300 ETE: BLK TIME: 7,0 FLT TIME: SPONSOR ORG: NOAA PROGRAM: STERAO -A PURPOSE : ORD PERSONNEL FENNEDYL AC SYS ENG ROLES CP FENUL DATA SYS MEMILLANL FORAL NAY RADAR FE TORREY BT/ODW RADIO SANS SOUCT -CLD PHYS FD WHITE /PARRISH " DOPPLER PARTICIPATING SCIENTIST/VISITORS/0A0 LAST, FIRST NAME ACTIVITY ON A/C REFILIATION HUBLERI NOAA/AL/CERES STERAD-A GOLD AN, PSJOBON, T 11 NOAA/AC SHERIPAN, POWILLIANS, J 11 NOAA/A FRIED, A 11 NCAR RYERSON 11 NOAA/AC/CIRES MATEJKA, NOAA/NSSC NOAA / NISC PROPOSED/ACTUAL MISSION/REMARKS (RECCO, FIXES, STORM, PENET, NHOP #) Q9:26 1014,1 41N 103 50W

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	U.S. DEPT.	COMM. /NOA	A/0A0 - DA	TA SECTION	NORK FORM	NO.2 DROWF2				
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STERAO 96 TRANSIT

FLIGHT #3 H960622

TYPE OF DATA	SENSOR OR OPTION
INE	1
Accelerometer	1
Temperature probe	1
Altitude change option	PA
(for vertical winds)	
Static pressure	Rosemount fuselage
Dynamic pressure	Rosemount fuselage
Time source	Micro 99
Constants file	CO2963.CON

Notes:

There were ten time/data gaps: 2151:35 2151:40 2151:43 2151:50 0000:11 0000:20 0000:23 0000:30 0209:05 0209:20

The aircraft INE positions were renavigated with respect to GPS.

SPECIAL NOTE!!! Locations 80, 81 and 82 of record five on the standard tape contain vertical ground, vertical air and vertical speeds, respectively, computed using Dave Jorgensen's vertical wind algorithm. It is recommended that these values be used for vertical wind analysis.

Flight Meteorologist: Sean White: (813) 828-3310 ext. 3072

96062214

START: 1955:52 200501 END: 0300:50 030000

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020000	-0.5	+2.3
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Flightplan Thunderstorm Study (Scenario 3) Colorado

boundary-layer to free troposphere exchange

Doors closed: 13:50 Takeoff: 14:00 **Point A**, Buckley National Guard Field 39 42 N, 104 45W

Calibration at constant altitude, Observer calls Head direction of Pt. B (41 00N, 104 25W) at 15 kft MSL

Pt. A to Pt. B 150 km, ~ 19 min, (Climb < 10 min)

Stay at constant altitude for calibration, observer calls $12 \times 14:00-14:25 \quad 2020 - 2037$

Before the Storm

After Calibration descend to 1500 ft AGL along the way to Pt. C

Horizontal Survey within the PBL

Pt. B to Pt. C (40 10N, 105 00W), to Pt. D (40 10N, 103 30W), to Pt. E (40 30N, 104 10W)14:25-15:10205522113221232

Racetrack Profile up to 21 kft near Pt. E

==> Take first set of HC can samples at 1500 ft AGL, (3 kft, 17 kft, 21 kft MSL head east for 1 min 50 sec and climb at 1500 ft/min, reverse direction, head west for 1 min 50 sec, and continue climb etc up to 21 kft MSL) 15:10-15:20

Head direction of Pt. F (40 10N, 104 25W) Pt. E to Pt. F 60 km, 9 min 15:20-15:30

Super Cell Scenario

IF during any part of the flight Operation Center identifies an isolated storm, fly U-Pattern (S, W, N sides) around the storm at max altitude, 21 kft MSL, 17 kft MSL, 13 kft MSL, and 1500 ft AGL.

The sides of the boxes have to be long enough for the radar characterization of the wind field and the distance from the storm is determined by the requirements of the doppler radar. The exact location of the boxes at a given altitude will depend on the movement of the storm cell.

Should the time that is required to fly this pattern be too long for the anticipated flight duration, limit the number of constant altitude boxes.

This pattern should satisfy both the radar and chemistry objectives of STERAO.

Squall Line Scenario

Concentrate on edge of squall line closest to CHILL.

L-pattern on the southern (south-eastern) and the western side of the squall line.

I) Radar leg :	parallel to the line of storm cells at 2000 ft AGL (default) from Bsw
	to Bse.
II) Radar leg :	return from Bse to Bsw at 500 ft AGL (default).
×.	Max length of these legs 15 min (~90 km).
III) Radar leg :	perpendicular to the line of storms at the western edge of the
	squall line, Bsw to Bnw at 500 ft AGL (default). Length of this leg
	is determined by radar requirement.
IV) Chemistry :	Return to Bse; either directly under the storm in a direct line from
	Bnw to Bse or parallel to the radar legs, but in closer proximity to
	the active storm cells at 500 ft AGL.

Repeat this pattern II)-IV), if radar observations call for it. Adjust altitude if radio communication with CHILL require it.

V) Racetrack profile ahead of the squall line (in clear air) up to max altitude.

If there is an anvil within the altitude range of the P3, descend to that altitude VI) and attempt to approach the anvil by flying short legs (< 5 min) parallel to the line in progressively closer proximity to the storm (pilots disgression).

VII) Fly to the back side of the squall line around the storm at max altitude. Attempt again to approach the anvil region.

VIII) Racetrack profile behind the squall line.

IX) Radar and chemistry leg behind the squall line from Bne to Bnw.

If the radar indicates an anvil that is within the altitude range of the P3, attempt to transect part of the anvil if it is feasible and safe. Start these transect away from the cloud and attempt to move closer on subsequent transects.

Hydrocarbon Profile between Pt. F and Pt. G

At 15:35 head direction of Pt. G (41 00N, 104 25W)

Hydrocarbon profile between Pt. F and Pt. G at 21 kft, 17 kft, 13 kft MSL, and 1500 ft AGL

Start HC-leg at : 05, :20, :35, :50 so that the HC sample will be centered along this leg. Change altitude approximately 3 min after HC sample ends (**Observer calls**) 15:35-16:35

If at any point during this leg the **Observation Center** at the CHILL Radar site should indentify a target storm, switch to the **Active Storm Phase.**

During the Active Storm Phase

Characterize Airflow with Radar and Characterize In-Flow Region of Cloud

Cloud identification and flight coordinates will be given by CHILL Radar.

Radar Legs:

Straight lines parallel to and S to SW of the storm approximately 12 km away from the perimeter of the storm, orientation parallel to storm track at 7kft MSL (within the PBL, this should also be the inflow region of the storm)

until dissipation of storm.

Alternatives:	I) Straight leg parallel to the storm track.
===>	II) Fly L-pattern (south and west side of the storm). See Squall Line
	Scenario
===>	II) Repeat radar flight pattern at different altitude.
	(7kft, 16 kft, and > 22 kft MSL).

16:35-18:35

After Dissipation of the Storm:

Characterization of PBL after Storm Dissipation: Cross below location of the latest radar echo for 15 min (~100 km) 18:35-18:50

Characterization of Vertical Profile after Storm Dissipation **Race track Profile** at location of the latest radar echo of the storm, descend to 500 ft AGL and then climb to max altitude at 1500 ft/min Alternatives:

I) 16:35-17:50: If there is no appropriate storm identified in the study area, start HC profile between Pt. F (40 10N, 104 25W), and Pt. AH (40 10N, 103 25 W) at 1500 ft AGL, 13 kft MSL, 17 kft MSL, 21 kft MSL, 25 kft MSL.

II) 17:50-19:05: start HC profile between Pt. AH (40 10N, 103 25W) and AI (41 00N, 103 25W) at 25 kft MSL, 21 kft MSL, 17 kft MSL, 13 kft MSL, 1500 ft AGL

Climb to 12 kft MSL for calibration, 30 min

Return to Buckley

The detailed characterization of the 1 deg lat by 1 deg long box will hopefully help to pick the most appropriate orientation for the HC profiles for subsequent flights. It also allows to fall back into the thunderstorm mode at any time if appropriate storms should develop later on.



MDT (hrs)

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==> Take second set of HC can samples at 1500 ft AGL, 13 kft, 17 kft, 21 kft MSL 18:50-19:20

HC profile after the storm

North-South leg across the center of the last radar echo, the length of the HC-leg is approximately 100 km (not quite 1 deg latitude) 2/52 $0.135 \neq 0.115 \neq$

Climb to 12 kft MSL for calibration (Start NOy calibration, and stay at altitude for calibration 20:35-20:50, Observer calls)

Head direction of Buckley (39 42N, 104 45W) Pt. A ~ 20 min

Return to Buckley at ~21:00

 Pt. A
 (39 42N, 104 45W)

 Pt. B
 (41 00N, 104 25W)

 Pt. C
 (40 10N, 105 00W)

 Pt. D
 (40 10N, 103 30W)

 Pt. E
 (40 30N, 104 10W)

 Pt. F
 (40 10N, 104 25W)

 Pt. G
 (41 00N, 104 25W)

other points will be determined during the flight......

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